

DATE	TOTAL TIME IN	TOTAL TIME SINCE	TACH OR RECORDING METER	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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48 Globe Ave., Texarkana, AR 71854

9-Octoberber-2011 A/C T.T. 8899.0 Tach  
 N6225C, Cessna T210N S/N 21063837  
 TSIO-520-R, S/N 278796-R, Engine TT 1440.7 SMOH 0.0

Page 1 of 1

C/W AD 71-09-07 R1, by replacement and pressure test of the exhaust manifold, no defects noted, next due @ 8999.0 T.T. (100 hours)  
 Note: Above AD Inspection requirements extended from 50 to 100 hours per AMOC when Knisley Welding P/N K1250860-203 installed.  
 C/W AD 93-5-6, P/C/W on 8-1-2000 @ 7917.7 hours, next due @ 9917.7 T.T. (2000 hours).  
 C/W AD 84-26-02, by replacement of paper induction air filter, filter next due for replacement @ 9399.0 T.T. (500 hours).  
 C/W AD 2008-10-2, by inspection of alternate static air source selector valve, no defects noted.  
 C/W AD 11-10-09, by inspection of seat track and seat pins, next due @ 8999.0 (100 Hours) or 10-2012(12 months).  
 ELT checked as per FAR 91.207 and found to operate properly, R&R ELT Battery w/new battery, next due for replacement 10/2013.  
 C/W Oxygen Cylinder Hydrostatic Test, next due Sep/2014 (3 years).

1. Removed and replaced engine assembly and all accessories with overhauled engine and accessories (engine overhauled by RAM Aircraft, TSIO-520-R, S/N 278296-R, for details see engine log book). Note: Fuel injection system adjusted as per TCM SID97-3E.
2. Engine equipped with Slick P/N 6320 pressurized magnetos, for details see FAA Form 337 this date.
2. Repaired and painted engine baffling and replaced baffle seals as needed, repaired fuel drain lines, and heat shields etc. as needed.
3. Removed and replaced Throttle, Mixture and Propeller control cables with new cable assemblies.
4. Removed and replaced complete exhaust system with new Knisley Welding modified exhaust system, as per STC Number SA5012NM, and SE5009NM, P/N 1250860MA, (Inc. P/N K1250860-203 heat exchanger) see FAA form 337 this date for details.
5. Removed and replaced all engine compartment fuel and oil hoses with new RAM Supplied hoses (Kit P/N 1449-4).
6. Removed and replaced all engine shock mounts with new mounts.
7. Removed and replaced main ship battery with new battery (Gill P/N G242).
8. Removed and replaced propeller governor with overhauled governor, P/N C290D4U/T2.
9. Removed and installed propeller after overhaul (Overhauled by Memphis Propeller Service).
10. Removed and replaced throttle position to gear warning switch cable with new cable.
11. Removed and replaced fuel strainer cable with new cable.
12. Removed and replaced engine to airframe ground strap with new strap.
13. Removed and replaced engine induction air filter assembly with new filter assembly (Note: AD 84-26-02 Applies).
14. Removed and replaced alternate air door magnetic latch with new latch assembly.
15. Removed and replaced vacuum pump with overhauled vacuum pump (overhauled by Holly Aero).
16. Removed and replaced all main and nose landing gear flexible hydraulic hoses with new hoses (supplied from Cessna).
17. Serviced landing gear, wheel bearings, fluid reservoirs, and checked fuel screen.
18. Removed and replaced LH and RH, aft nose gear door hinges and hinge pins with new hinges and pins.
19. Performed landing gear warning horn and safety switch operational test & found the warning horn & safety switch to operate properly.
20. Removed and replaced LH and RH brake linings with new linings and resealed RH and LH brake calipers and bled brake system.
21. Checked all control cable tensions and found within limits.
22. Removed and replaced nose tire and tube with new Goodyear Flight Custom III tire and new tube.
23. Removed and replaced LH and RH main tires and tubes with new Goodyear Flight Custom II tires and new tubes.
24. Removed and replaced oxygen cylinder with serviceable cylinder (P/N 176307-76, S/N K600765/54). Cylinder was re-certified (hydrostatic tested) by Shoreline Marine, dba Safetech, see W/O number XD11030 dated 9/16/2011 for details.
25. Removed and replaced pilot inboard and outboard and co-pilot inboard and outboard seat rails with new rails.
26. Removed and replaced LH and RH inner and outer cabin side windows and aft cabin window with new windows.

I certify that this Aircraft has been inspected in accordance with a 100 Hour Inspection and found to be in airworthy condition with respect to the work accomplished. \_\_\_\_\_END\_\_\_\_\_

  
 Jerry M Garrett

A&P 2688220

Date 9-10-2011


SUB-TOTALS this page

TOTALS—Carry forward to next page

Page No. \_\_\_\_\_

**LOG BOOK ENTRY**

**E.L.T. BATTERY**

**REPLACEMENT DATE** OCT/2013

PTION OF WORK PERFORMED—  
IFICATE NO. OF PERSON PERFORMING WORK

DATE	TOTAL TIME IN SERVICE	TOTAL TIME SINCE OVERHAUL



48 Globe Ave., Texarkana, AR 71854

9-Octoberber-2011 A/C T.T. 8899.0 Tach  
N6225C, Cessna T210N S/N 21063837  
TSIO-520-R, S/N 278796-R, Engine TT 1440.7 SMOH 0.0

Page 1 of 1

I certify that this Aircraft has been inspected in accordance with an Annual Inspection and was found to be in airworthy condition.  
-----END-----

James B. West  
James B. West

A&P 3087142 IA

Date 10-09-2011

**Mena Aircraft Painting**  
120 Aviation Lane Mena Ar 71953  
Ph#479-243-9995 Fax#479-243-9996

A/C: CESSNA 210 N#6225C SR#21063837

Date 12/1/11

THE ABOVE MENTIONED AIRCRAFT WAS HERE FOR PAINT.  
THE AIRCRAFT WAS STRIPPED, PRIMED, AND REPAINTED WITH SHERWIN WILLIAM PRODUCTS.  
BASE COLOR: MATTERHORN WHITE #570-535  
TRIM COLORS: ACRY GLO, RUBY #10701, HARVEST GOLD 10653, REGIMENT BLUE 07447.  
ALL REQUIRED FLIGHT CONTROLS WERE REMOVED, CHECKED FOR BALANCE AND REINSTALLED  
IN ACCORDANCE WITH CESSNA FLIGHT MANUAL.  
FILED UNDER FILE#753

AUTHORIZED SIGNATURE: DOUG BLAIR

Doug Blair

A#451041859

**Blair-Air Interiors**  
120 Aviation Lane Mena Ar 71953  
Ph#479-243-9995 Fax#479-243-9996

A/C: CESSNA 210 N#6225C SR#210

Date: 12/1/11 TACH/HRS

THE ABOVE MENTIONED AIRCRAFT WAS HERE FOR INTERIOR.  
HEADLINER: VASI WAYPOINT PAMPAS WP830  
LEATHER: DANI WET SAND C2106/DANIPALE TUNDRA C2172  
CARPET: AIRABODE DAVINCI BEAVER 2604  
SEATBELTS: C & M MARINE AVIATION  
ALL MATERIALS MEET FAR 25.853 REQUIRMENTS  
FILED UNDER FILE #753

AUTHORIZED SIGNATURE: DOUG BLAIR

Doug Blair

A#451041859





DATE	TOTAL TIME IN SERVICE	TOTAL TIME SINCE OVERHAUL	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
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48 Globe Ave., Texarkana, AR 71854

9-Octoberber-2011 A/C T.T. 8899.0 Tach  
 N6225C, Cessna T210N S/N 21063837  
 TSIO-520-R, S/N 278796-R, Engine TT 1440.7 SMOH 0.0

Page 1 of 1

1. Removed and replaced engine assembly and all accessories with overhauled engine and accessories including turbo charger, turbo oil supply and return check valves, turbo controller, wastegate, alternator, starter, fuel injection system etc. (engine overhauled by RAM Aircraft, for details see engine log book entry dated May 27<sup>th</sup>, 2011).
2. Engine equipped with Slick P/N 6320 pressurized magnetos, for additional details see FAA Form 337 this date.
3. Repaired and painted engine baffling and replaced baffle seals as needed, repaired fuel drain lines, and heat shields etc. as needed.
4. Removed and replaced complete exhaust system with new Knisley Welding modified exhaust system, as per STC Number SA5012NM, and SE5009NM, P/N 1250860MA, (Inc. P/N K1250860-203 heat exchanger) see FAA form 337 this date for details.
5. Removed and replaced all engine compartment fuel and oil hoses with new RAM Supplied hoses (Kit P/N 1449-4).
6. Removed and replaced all engine shock mounts with new mounts.
7. Removed and replaced propeller governor with overhauled governor, P/N C290D4U/T2.
8. Removed and replaced engine induction air filter assembly with new filter assembly (Note: AD 84-26-02 Applies).
9. Removed and replaced alternate air door magnetic latch with new latch assembly.
10. Removed and replaced vacuum pump with overhauled vacuum pump (overhauled by Holly Aero).
11. Engine was serviced and pre-oiled prior to initial start with Phillips 20W-50 Type "M" engine oil, engine fuel system was then adjusted as per TCM SID97-3E. Engine was then checked for oil, fuel and exhaust leaks, none noted. Engine operation normal at all power settings.

I certify that this Engine has been inspected in accordance with a **100 Hour** Inspection and was determined to be in airworthy condition.  
 -----END-----

*Jerry M. Garrett*  
 \_\_\_\_\_  
 Jerry M. Garrett                      A&P 2678534

\_\_\_\_\_ Date 10-09-2011



48 Globe Ave., Texarkana, AR 71854

9-Octoberber-2011 A/C T.T. 8899.0 Tach  
 N6225C, Cessna T210N S/N 21063837  
 TSIO-520-R, S/N 278796-R, Engine TT 1440.7 SMOH 0.0

Page 1 of 1

I certify that this Engine has been inspected in accordance with an **Annual** Inspection and was found to be in airworthy condition.  
 -----END-----

*James B. West*  
 \_\_\_\_\_  
 James B. West                      A&P 3087142 IA

\_\_\_\_\_ Date 10-09-2011


SUB-TOTALS this page

TOTALS—Carry forward to next page



# AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

1. Approving Nat. of Aviation Authority: <b>Country: FAA/UNITED STATES</b>		3. System Tracking Ref. No.: <b>RMA 36692075</b>
Organization: <b>GARMIN International 1200 E 151st Olathe, KS 66062</b>	Certificate No.: <b>G6XR582Y</b>	5. Work Order Contract/Invoice Number: <b>RMA 36692075</b>

Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1.	GNS430	011-00280-10	N/A	1	97101414	REPAIRED

**13. REMARKS:** Confirmed the return issue. Repaired CDU to correct. Updated software to the latest version. Aligned unit for optimum performance. Unit passes all tests and is operating within normal limits. Unit acquired satellites. The Main and GPS software's are at the latest revision. This unit complies with Service Bulletins No. 9906, 9907, 9908, 0003, 0004, 0007, 0009, 0011, 0014, 0016, 0020, 0021, 0022, 0105, 0107, 0116, 0201, 0204, 0205, 0216, 0218, 0334, 0308, 0309, 0502, 0515, 0521, 0706, and 0844. It is the responsibility of the Installer to verify that the installation complies with the previous listed Service Bulletins. If this unit is part of a dual unit installation Garmin recommends that both units have the same software levels. It is the responsibility of the installer to assess installation compatibility.

This unit complies with MOD 1 per Garmin's Service Bulletin No. 0019  
 This unit complies with MOD 3 per Garmin's Service Bulletin No. 0203B

This "Certifies that the work specified in block 12/13 was carried out in accordance with EASA Part-145 and in respect to that work the component is ready for release to service under EASA Part-145 Approval Number: EASA.145.5534". The work that was performed on this unit was done to meet the requirements of all sections of the maintenance manual part number 190-00364-00 Rev- AF, Revision Date 9/9/2011.

Thank You for Choosing Garmin! If you have any questions or concerns please do not hesitate to contact our customer service department at 1-800-800-1020. Please view Garmin's web site at [www.garmin.com](http://www.garmin.com) for any update or product information.

All repaired or replaced GARMIN units have a 90-day warranty or the continuation of the original factory warranty from the original purchase date, whichever is longer.

<p>14. Certifies the items identified above were manufactured in conformity to:</p> <p><input type="checkbox"/> Approved design data and are in condition for safe operation</p> <p><input type="checkbox"/> Non-approved design data specified in block 13</p>	<p>19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service      <input checked="" type="checkbox"/> Other regulations specified in Block 13</p> <p>Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.</p>		
<p>15. Authorized Signature: <b>N / A</b></p>	<p>16. Approval/Authorization No.: <b>N / A</b></p>	<p>20. Authorized Signature: <i>Stephen E. Abels</i></p>	<p>21. Approval/Certificate No.: <b>G6XR582Y</b></p>
<p>17. Name (Typed or Printed): <b>N / A</b></p>	<p>18. Date (m/d/y): <b>N / A</b></p>	<p>22. Name (Typed or Printed): <b>Stephen E. Abels</b></p>	<p>23. Date (m/d/y): <b>Dec 09 2011</b></p>

### User/Installer Responsibilities

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

Statements in block 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

DATE	TOTAL TIME IN SERVICE	TOTAL TIME SINCE OVERHAUL	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
<b>TOTALS</b> brought forward from previous page				
1-3-07	Tach time: 8684.8	Total time Propeller: 1154.2		Inspected prop in accordance with applicable sections of FAR 43 appendix D. I certify this propeller has been inspected in accordance with an annual inspection and was found to be in airworthy condition. David Bogan IA 455216543 <i>David Bogan</i>
1-8-08	N6225C	Tach time: 8794.3	Propeller total time: 1263.7	Inspected propeller in accordance with applicable sections of FAR 43 appendix D. I certify this propeller has been inspected in accordance with an annual inspection and was found to be in airworthy condition. David Bogan IA 455216543. <i>David Bogan</i>
1-June-2011	TACH 8899.7	HOBBS 1807.8		Removed FOR OVERHAUL. AP 2688710 <i>Jan [Signature]</i>
<b>THIS PROPELLER/COMPONENT HAS</b>				

OMB Control No. 2120-0018  
09/30/2007

1. Approving National Aviation Authority / Country: FAA/UNITED STATES		<b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: 061011	
4. Organization Name and Address: MEMPHIS PROPELLER SERVICE, INC. 11098 Willow Ridge Drive - Olive Branch, Mississippi 38654				PI4R441M		5. Work Order/Contract/Invoice No. 18580	
6. Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial / Batch Number:	12. Status / Work:	
1	Propeller	PHC-J3YF-1RF	NA	1	FP818A	Overhaul	
13. Remarks: Overhauled Propeller Assembly In Accordance with FAA Approved Data. Full Details of Work Carried Out Per Work Order #18580. TSN: 1369.1 TSO: 0.0 Blade Model: F7663D-2Q S/N: J06491, J16496, J16499							
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.			19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.				
15. Authorized Signature:		16. Approval/Authorization No.:		20. Approval/Certificate No.:		21. Approval/Certificate No.:	
				PI4R441M			
17. Name (Typed or Printed):		18. Date (m/d/y):		22. Name (Typed or Printed):		23. Date (m/d/y):	
		00/00/00		Trong Vu, Chief Inspector		Jun 10, 2011	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							





United States of America  
Department of Transportation — Federal Aviation Administration  
**Supplemental Type Certificate**

*Number* SA4111SW

*This certificate, issued to:* RAM Aircraft Corporation  
P. O. Box 5219  
Waco, Texas 76708

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 3 of the Civil Air Regulations.*

*Original Product — Type Certificate Number:* 3A21, A4CE

*Make:* Cessna

*Model:* T210, T210G, T210H, T210J, T210K, T210L, T210M, T210N, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F, TU206G, TP206A, TP206B, TP206C, TP206D, TP206E

*Description of Type Design Change:*

Installation of Hartzell Model PHC-J3YF-1RF/F7663-2Q, or F7663D-2Q Propeller according to Drawing 21004, Revision E dated 6/3/85, or later FAA approved revision.

*Limitations and Conditions:*

See Continuation Sheet, pages 2 and 3, of this STC.

*This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.*

*Date of application:* March 20, 1980

*Date issued:*

*Date of issuance:* April 10, 1980

*Date amended:*

9/3/81; 9/14/81; 10/7/81;

4/25/82; 8/16/85 Rev. 5



*By direction of the Administrator*

*Don P. Watson* (Signature)

Don P. Watson  
Manager, Aircraft Certification Division  
Southwest Region

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

7/28/06

N6225C

INSTALLED AERO SAFE CORPORATION  
STABBY VACUUM SYSTEM IN ACCORDANCE WITH  
STC SA 4626SW. INSTALLED IAW DRAWING LIST  
820200, REV. B, DATED 4/28/82 OR LATER AD-  
PROVED FAA REVISION.

ALL WORK ACCOMPLISHED MEETS REQUIREMENTS  
OF CESSNA T210 MAINTENANCE MANUAL.

ELECTRICAL LOAD ANALYSIS SHOWS  
CONTINUAL LOAD DOES NOT EXCEED 80% OF GENERATOR  
OUTPUT.

WIRING CONFORMS TO AC 43.13-1B CHAPT.  
11; SECTION 8 - WIRING INSTALLATION REQUIREMENTS.

WEIGHT + BALANCE CHANGE NOTED IN A/C RECORDS.  
EAD

Additional Sheets Are Attached

Department of Transportation -- Federal Aviation Administration  
**Supplemental Type Certificate**

*Number* SA966CE

*This certificate, issued to* Horton STOL-Craft  
 Wellington Municipal Airport  
 Wellington, KS 67152

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified herein meets the airworthiness requirements of Part 3 of the Civil Air Regulations effective May 15, 1956, and Paragraph 3.112 as amended October 1, 1959.*

*Original Product -- Type Certificate Number:* 3A21  
*Make:* Cessna  
*Model:* 210G, T210G, 210H, T210H, 210J, T210J, 210K, T210K, 210L, T210L, 210M, T210M, 210N, T210N and P210N

*Description of Type Design Change:*

Installation of leading edge cuffs, droop tips, aileron gap seals, and stall fences, per STOL-Craft Drawing List No. 3 with FAA approval date of November 8, 1973, or later; and Drawing List No. 3 with an FAA approval date of April 30, 1986, for 210L, T210L, 210M, T210M, 210N and T210N modified by STC SA760NW; and Drawing List No. 3 with an FAA  
 (See Continuation Sheet)

*Limitations and Conditions:* 1. Airplane Flight Manual Supplement dated April 25, 1980, is required for 210N/T210N airplanes, Serial Number 21062955 and up, and Airplane Flight Manual Supplement dated April 20, 1982, is required for P210N airplanes, Serial Number P21000151 and up when this modification is installed.  
 2. Airplane Flight Manual Supplement dated April 25, 1980, revised to October 5, 1988, is required equipment for 210/T210 airplanes  
 (See Continuation Sheet)

*This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.*

*Date of application:* October 23, 1973

*Date reissued:* May 25, 1977

*Date of issuance:* October 23, 1973

*Date amended:* 5/25/78, 4/25/80, 2/8/82, 4/20/82, 4/30/88, 10/6/88,  
*By direction of the Administrator* 4/17/92



*Ronald K. Ratlgebe*  
 for Lawrence A. Herron, <sup>(Signature)</sup> Manager  
 Wichita Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



Department of Transportation—Federal Aviation Administration  
**Supplemental Type Certificate**  
(Continuation Sheet)

*Number* SA966CE

Description of Type Design Change, Continued:

approval date of October 6, 1988, for 210N, T210N and P210N is required for installation of de-icer boots over the leading edge cuffs.

Limitations and Conditions, Continued:

2. (Continued) previously equipped with de-icer boots.

3. Airplane Flight Manual Supplement dated April 20, 1982, revised to October 4, 1988, is required equipment for P210N airplanes previously equipped with de-icer boots.

4. 210N, P210N, and T210N airplanes are eligible for installation of de-icer boots over the leading edge cuffs providing the airplane has the factory installed provisions for de-icing.

5. T210N airplanes previously modified by STC SA760NW are eligible for this modification.

6. 210K, T210K, 210L, T210L, 210M and T210M airplanes previously modified by STC SA1202CE are eligible for this modification.

7. This approval should not be extended to other specific airplanes of these models on which other previously approved modifications are incorporated, unless it is determined that the interrelationship between this change and any of those other previously approved modifications will introduce no adverse effect upon the airworthiness of those airplanes.

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*Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.*



## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020
<b>For FAA Use Only</b>
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

<b>1. Aircraft</b>	Make <b>Cessna</b>	Model <b>T210N</b>
	Serial No. <b>21063837</b>	Nationality and Registration Mark <b>N6225C</b>
<b>2. Owner</b>	Name (As shown on registration certificate) <b>EZ Aviation, Inc.</b>	Address (As shown on registration certificate) <b>16950 Dallas Parkway Suite 105 Dallas, TX 75248</b>

### 3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				<b>XX</b>
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

<b>A. Agency's Name and Address</b> <b>Alvin Janzen</b> <b>P.O. Box 356</b> <b>Oxford, KS 67119</b>	<b>B. Kind of Agency</b> <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	<b>C. Certificate No.</b> <b>A&amp;P 513402783</b>
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

<b>Date</b> <b>November 9, 1992</b>	<b>Signature of Authorized Individual</b> 
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### 7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

<b>3Y</b>	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
<b>Date of Approval or Rejection</b> <b>November 9, 1992</b>		Certificate or Designation No. <b>661870</b>	<b>Signature of Authorized Individual</b> 		



**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed De-ice Boots, deactivated switch and marked INOP.
2. Installed Horton STOL-Kit, kit serial number 4990, in accordance with drawing 2053 Sheet 1, approved by STC# SA966CE.
3. Flight Manuel Supplement placed in Flight Manuel.
4. Weight & Balance & Equipment list updated.

.....THE END.....